EC130B4 NORMAL CHECK LIST



ALLWAYS REFER TO AFM FOR MANUFACTURER PROCEDURES

| LIMITATIO | INTERNAL ONLY | MAXIMUM | 2427 KG |
|--------------------|----------------------|--------------------|-----------------------------|
| | WITH EXTERNAL LOAD | | 2800 KG |
| | Оп Ноок | | 1160 кд |
| SPEED (VNE) | Power On | Maximum | 155 KIAS – 3ктs/1000гт |
| , , | Power Off | | 125 KIAS – 3KTS/1000FT |
| | LH SLIDING DOOR OPEN | | 80 KIAS |
| | OPENING SLIDING DOOR | | 70 KIAS |
| | OTHER DOOR CONFIG | | REF RFM SECT 2 |
| | EXTERNAL LOAD | | 80 KIAS |
| ALTITUDE | | Махімим | 23,000 FT PA |
| TEMP | OAT | Мімімим | -40°c |
| | | Махімим | ISA +35°C LIMITED TO +50°C |
| MAIN ROTOR | Power On | STABILIZED FLIGHT | 375 то 405 грм |
| (NR) | | CAUTION HIGH | 405 to 430 rpm |
| | | CAUTION LOW | 320 то 375 крм |
| | Power Off | Махімим | 430 RPM |
| | | Мімімим | 320 RPM |
| | ROTOR BRAKE APPLY | Махімим | 170 RPM |
| FIRST LIMIT | Махімим | Continuous | 9.6 |
| INDICATOR | | TAKE OFF POWER | 9.6 то 10 |
| (FLI) | | TRANSIENT | 10.4 (5 SECONDS) |
| TORQUE | | Max Continuous | 92.7% |
| | | Max Take Off | 92.7% то 100% (0 то 40 ктѕ) |
| | | MAX TRANSIENT | 104% (5 SEC LIMIT) |
| FREE | Power ON | Мімімим | 349 RPM |
| TURBINE (NF) | | MAXIMUM CONTINUOUS | 349 то 418 грм |
| | | Max transient | 450 RPM (5 SEC LIMIT) |
| T4 | START & SHUT DOWN | Махімим | 750°C CONTINUOUS |
| | | | 865°C (10 SEC LIMIT) |
| | TAKE OFF | Махімим | 915°c |
| | In Flight | MAXIMUM CONTINUOUS | 849°c |
| GENERATOR | LOAD | Махімим | 150 Amp |
| | | VOLTAGE MAX | 31.5 V |
| SLOPE LIMITS | MAXIMUM | NOSE UP | 6° |
| | | Nose Down | 6° |
| | | SIDEWAYS | 8° |
| CREW | | | 1 LEFT SIDE |
| PERSONS CARRIED | | | 7 (INCLUDING PILOT) |
| FUEL | | Standard | 540 LITER |
| FUEL BURN | | Махімим | ~190 L/HOUR |

FLIGHT MANUAL EC130 B4



4.3 START UP

4.3.1 ENGINE PRESTART CHECK

NOTE

Copilot seat belts shall be fastened in all cases.

| 1. Rotor brake | RELEASED, | fully | forward. |
|----------------|-----------|-------|----------|

- - condition.
- LOCK-WIRED
- 5. Starting selector......OFF.
- 7. [EXT PWR BATT.] or [BAT/EPU]

[DIRECT BATT.] or [DCT BAT]
[GEN] or [GENE]ON.

- 8. ICS and GPS nav.system (if fitted)....... ON.

(night flight intended).

- 11. [W/LT TEST] or [W/LT TST] COMPLETE.
- 13. [SERVO/TEST] or [SERVO TST].....SERVO when depressed.
- 14. CWP lights CHECK :

| With battery pow | er GENE | PITOT | ENG P |
|------------------|---------|-------|---------|
| | FUEL P | HORN | MGB.P |
| | SERVO | HYDR | TWT GRP |

• With EPU power Same lights as above + BATT

| 15. VEMD | 3-data page DISPLAYED |
|------------------------------|---------------------------------------|
| | Vehicle page DISPLAYED |
| | Battery voltage > 22 V |
| 16. Control pedals | Freedom of travel, then NEUTRAL. |
| 17. Cyclic pitch | CENTER, friction adjusted. |
| 18. Collective pitch | LOCKED. |
| 19. Heating, demisting, | |
| air conditioning (if fitted) | OFF. |
| 17. Cyclic pitch | CENTER, friction adjusted. LOCKED. |

| 1.3.2 | ENGINE STARTING |
|-------|---|
| | 1. CWPCHECK GOV |
| | 2. [FUEL PUMP] or [FUEL P]ON |
| | 3. [A/COL LT]ON |
| | • After 30 s |
| | 4. Starting selectorON position 5. CheckNg increases |
| | . T4 remains below its limits |
| | . Rotor is turning at Ng \geq 25 % |
| | . Engine oil pressure increases |
| | • When $Ng \ge 67 \%$ |
| | 6. CWPCHECK: |
| | ENG P , MGB P , HYDR. |
| | SERVO, FUEL P |
| | 7. [PITOT] or [PITOTS]ON, PITOT |
| | 8. [FUEL PUMP] or [FUEL P]OFF |
| | 9. [SERVO/TEST] or [SERVO TST] SERVO when depressed. |
| | 10.Starting selector guardLOWER |
| | 11.All necessary systemsON - TESTED (Master avionic switch, lights, etc.) |
| | NOTE |
| | At Ng > 60 % the VEMD upper screen automatically switches to FLI display. |
| | • If EPU is used: 12. EPUDISCONNECT GENE, BATT |
| | Make sure EPU door closed. |
| | |

4.3.3 RUN-UP CHECK

- 1. Twist gripFLIGHT detent.
 - When $NR \ge 340 \text{ rpm}$
- 2. [HORN]ON, HORN

CHECK aural warning:

- . ON for NR \leq 360 rpm and
- . OFF for NR > 360 rpm.
- 3. NR indicationCHECK in lower green range.

4.3.4 CRANKING

The cranking procedure can be performed after an aborted start or for check or maintenance purposes.

Proceed as follows:

- 1. Start selectorOFF.
- 3. Ng......CHECK ≤ 10 %.
- 4. [FUEL PUMP] or [FUEL P]....ON.
- 6. [CRANK]OFF.
- [FUEL PUMP] or [FUEL P] OFF.

CAUTION

Do not crank the engine with the emergency fuel shutoff valve closed or with the fuel pump off as this could damage the engine high pressure fuel pump.

4.4 TAKEOFF

4.4.1 BEFORE TAKEOFF CHECK

- 2. Cyclic and collective frictions AS REQUIRED.
- 3. Landing light...... AS REQUIRED.
- 4. Temperatures and pressures......NORMAL RANGE.
- 5. Warning panel All lights OFF.

NOTE

Adjust collective and cyclic friction so that friction forces are felt by the pilot when moving the flight controls.

4.4.2 TAKEOFF CHECK AND PROCEDURE

CAUTION

Use of P2 air bleeds is forbidden above engine maximum continuous rating (Ng or T4).

- Gradually increase collective pitch to hover at 5 ft. Check engine and mechanical control instruments, no warning light.
- Increase airspeed with HIGE power until IAS = 40 kt, then begin to climb so as to clear 20 ft at IAS = 50 kt.

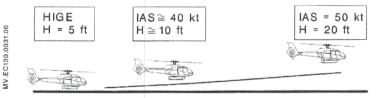


Figure 4-2: Takeoff Procedure

CAUTION

For safe operation, takeoff path should avoid HV diagram (refer to SECTION 5).

| 4.8 | ENGINE AND ROTOR SHUTDOWN |
|-----|--|
| | 1. Cyclic stick NEUTRAL |
| | 2. Collective pitchLOCK |
| | 3. Twist grip |
| | 4. Engine oil cooling |
| | 5. [PITOT] or [PITOTS], [HORN], Landing light OFF |
| | 6. Non-required systems, |
| | [MASTER AVIONICS] or [AVIONIC]OFF |
| | 7. Starting selector OFF position |
| | 8. [GEN] or [GENE] OFF |
| | For NR ≤ 140 RPM normal NR |
| | 170 RPM maximum NR (in strong wind operations) |
| | 9. Rotor brake APPLY |
| | |
| | • When rotor is stopped: |
| | 10. GPS navigation system (if fitted) OFF |
| | 11. [A/COL LT]OFF |
| | |
| | • BEFORE LEAVING HELICOPTER |
| | 12. VEMD |
| | . Operating time (counted from Ng > 60 % until Ng < 50 %) |
| | . Ng and Nf cycles |
| | and above 0). |
| | |
| | . Advisory messages of FAILURE or OVERLIMIT DETECTED |
| | 13. [DIRECT BATT.] or [DCT BAT], |
| | [EXT PWR BATT.] or [BAT/EPU] OFF |
| | 14. Pitot, air intake and exhaust covers, blade socks as required. |
| 1 | |

EC130B4 CHECKLIST

